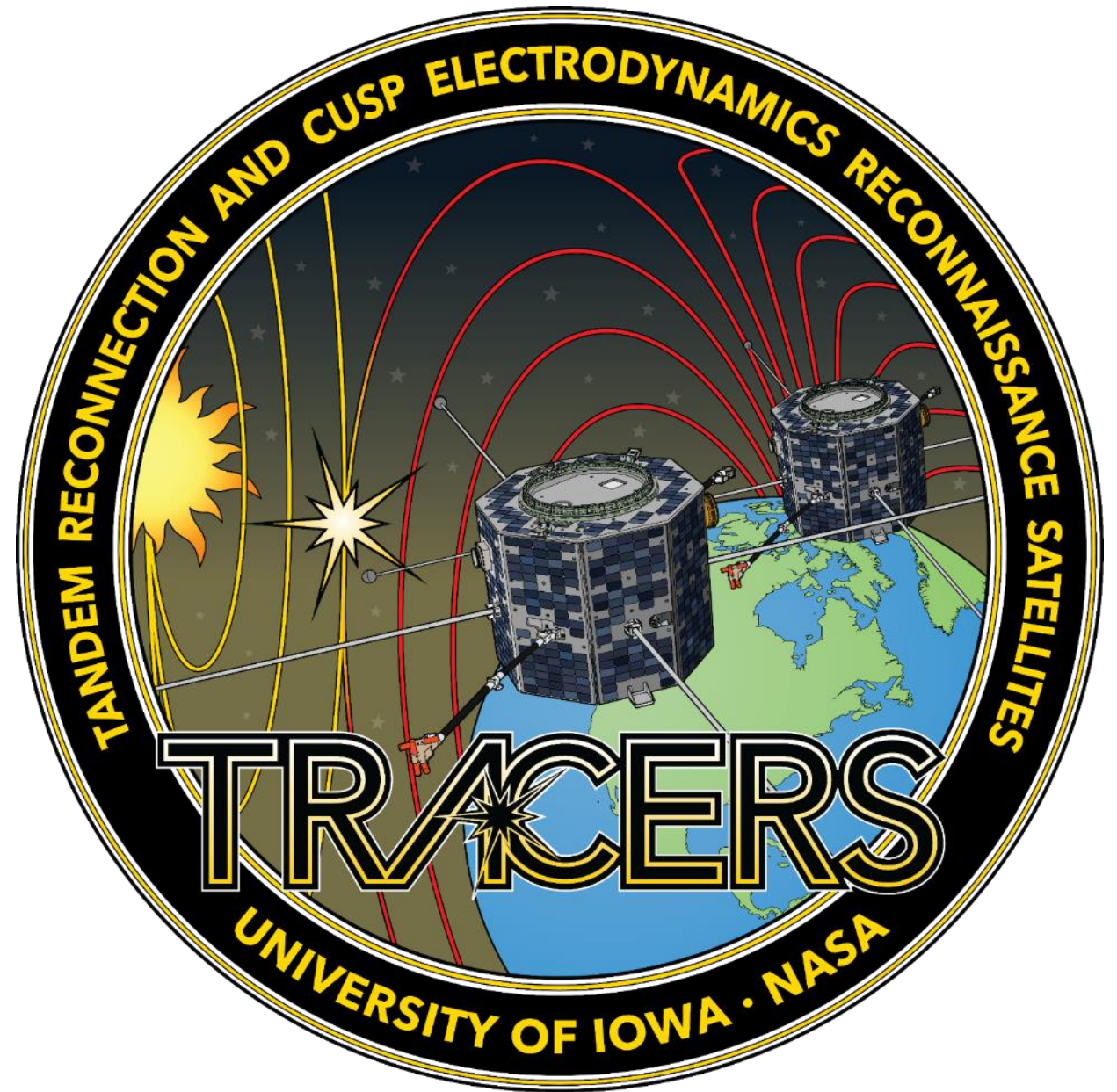

The upcoming Tandem Reconnection and Cusp Electrodynamics Reconnaissance Satellites (TRACERS) Small Explorers Mission

David Miles
University of Iowa
david-miles@uiowa.edu

DASP 2024, Edmonton, AB

February 21, 2024

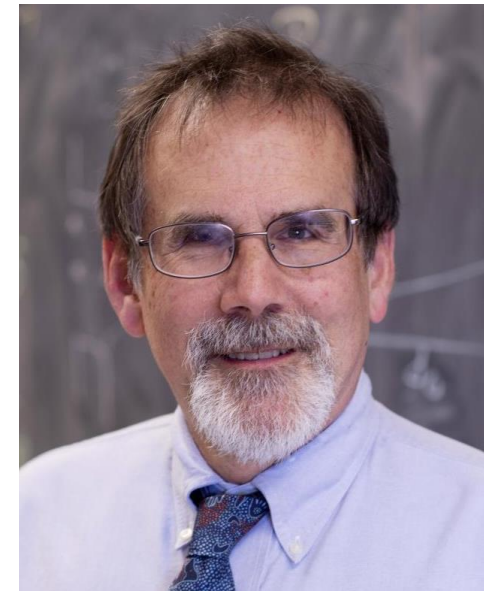


TRACERS

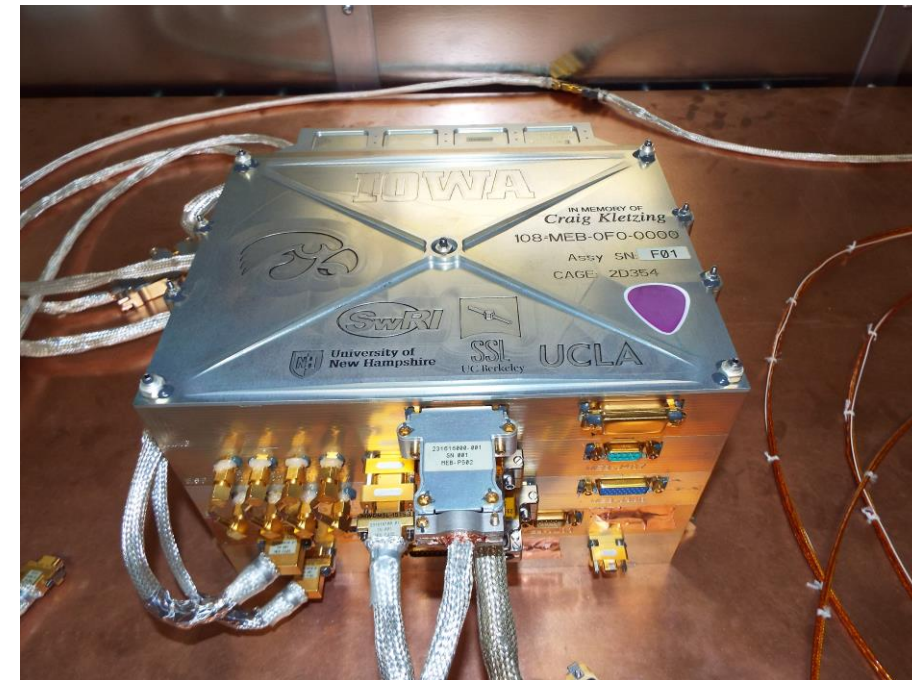
Department of Physics and Astronomy

Passing of TRACERS PI Prof. Craig Kletzing

- Prof. Craig Kletzing died peacefully at home on August 10, 2023.
- The University of Iowa will hold a celebration of Craig's career on June 15, 2024 immediately after a TRACERS Science Team Meeting.

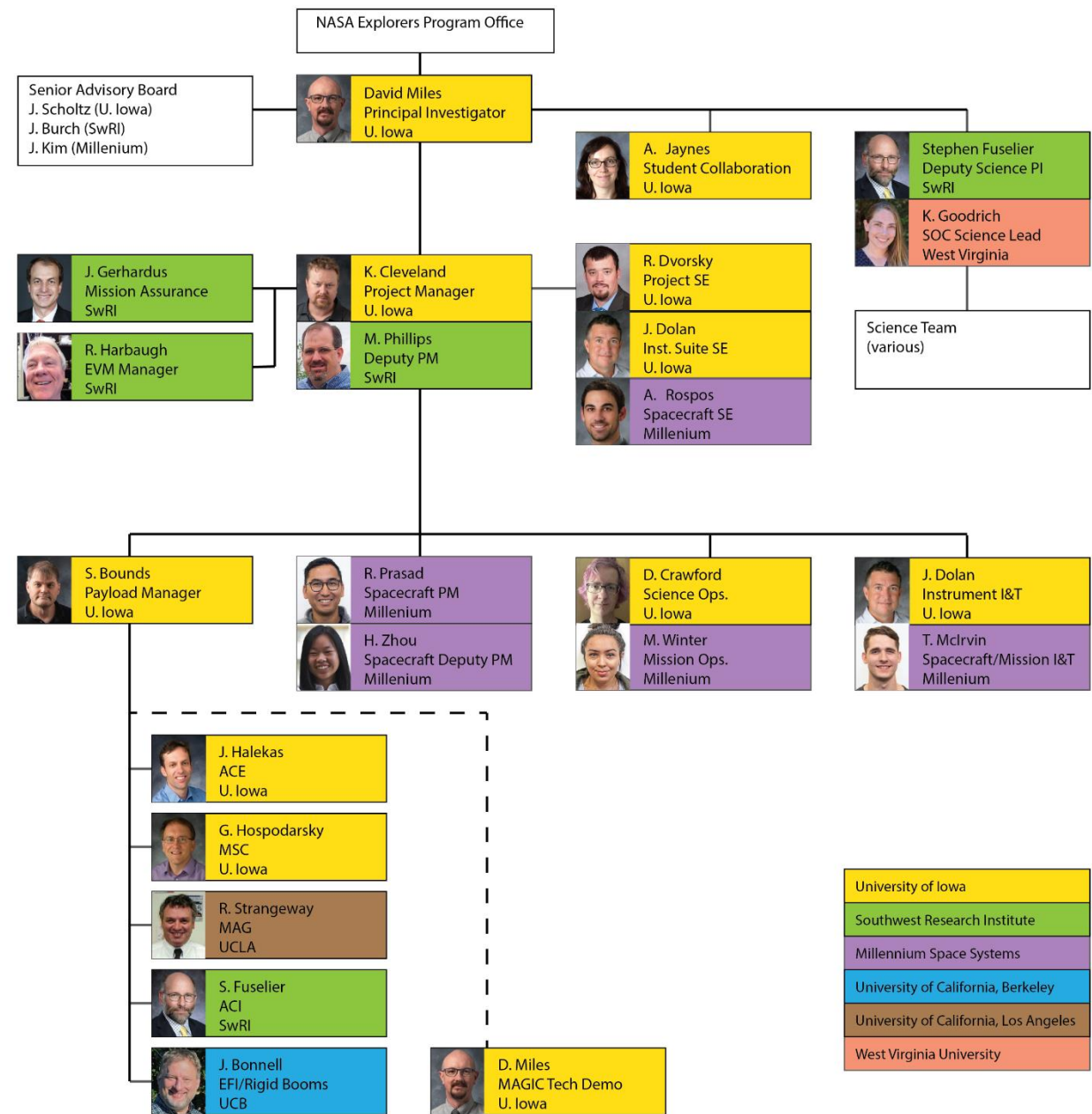


<https://now.uiowa.edu/news/2023/08/renowned-iowa-physics-professor-researcher-craig-kletzing-dies>



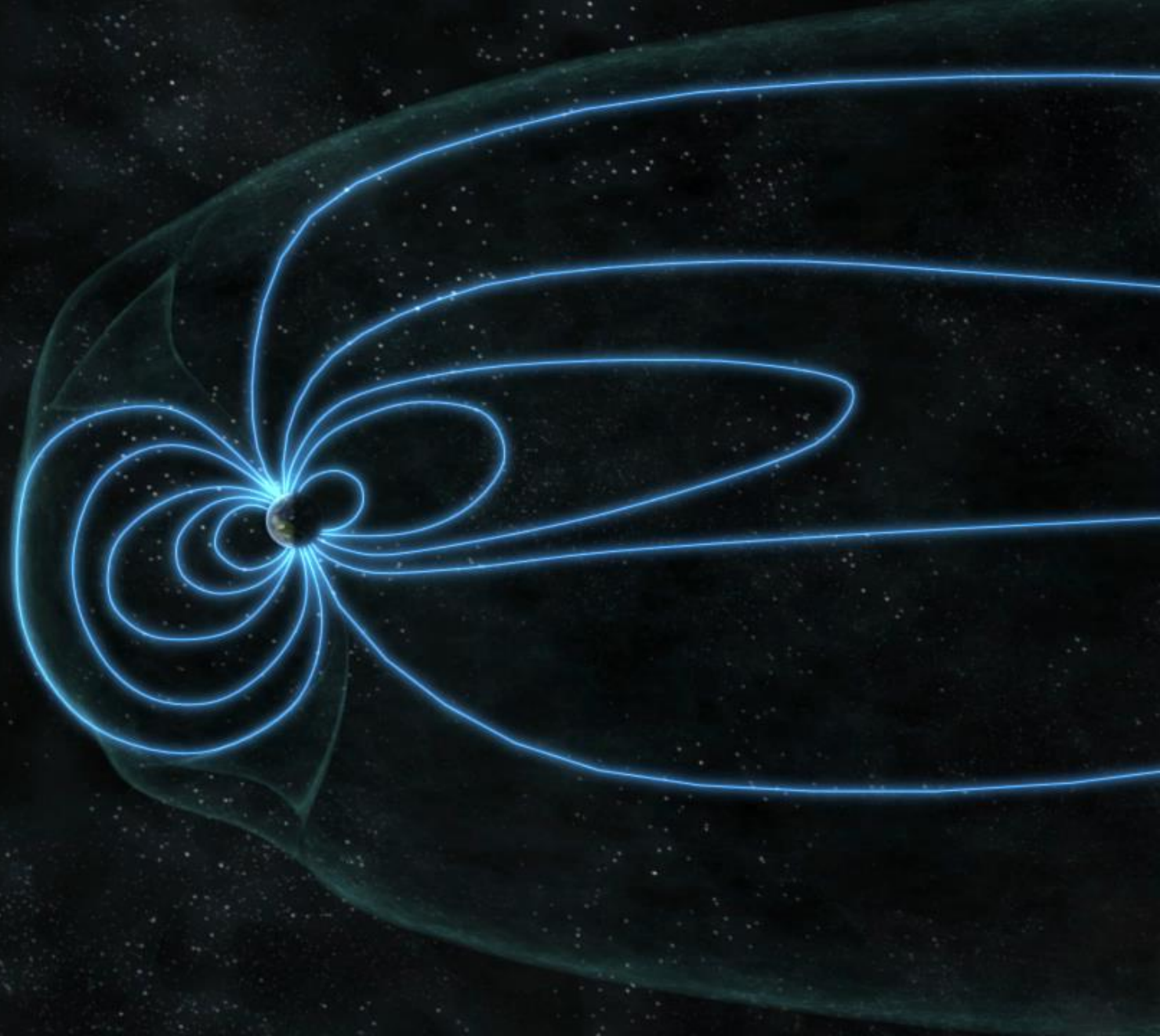
What is a SMEX Mission?

- NASA's Explorers Program has been around since the start of the space age - Explorer 1!
- The Principal Investigator (PI) and their team are responsible for everything but the launch.
- Small Explorers (SMEX) are cost-capped with focused science objectives.
- \$165 M PIMCC with a NASA-provided launch.



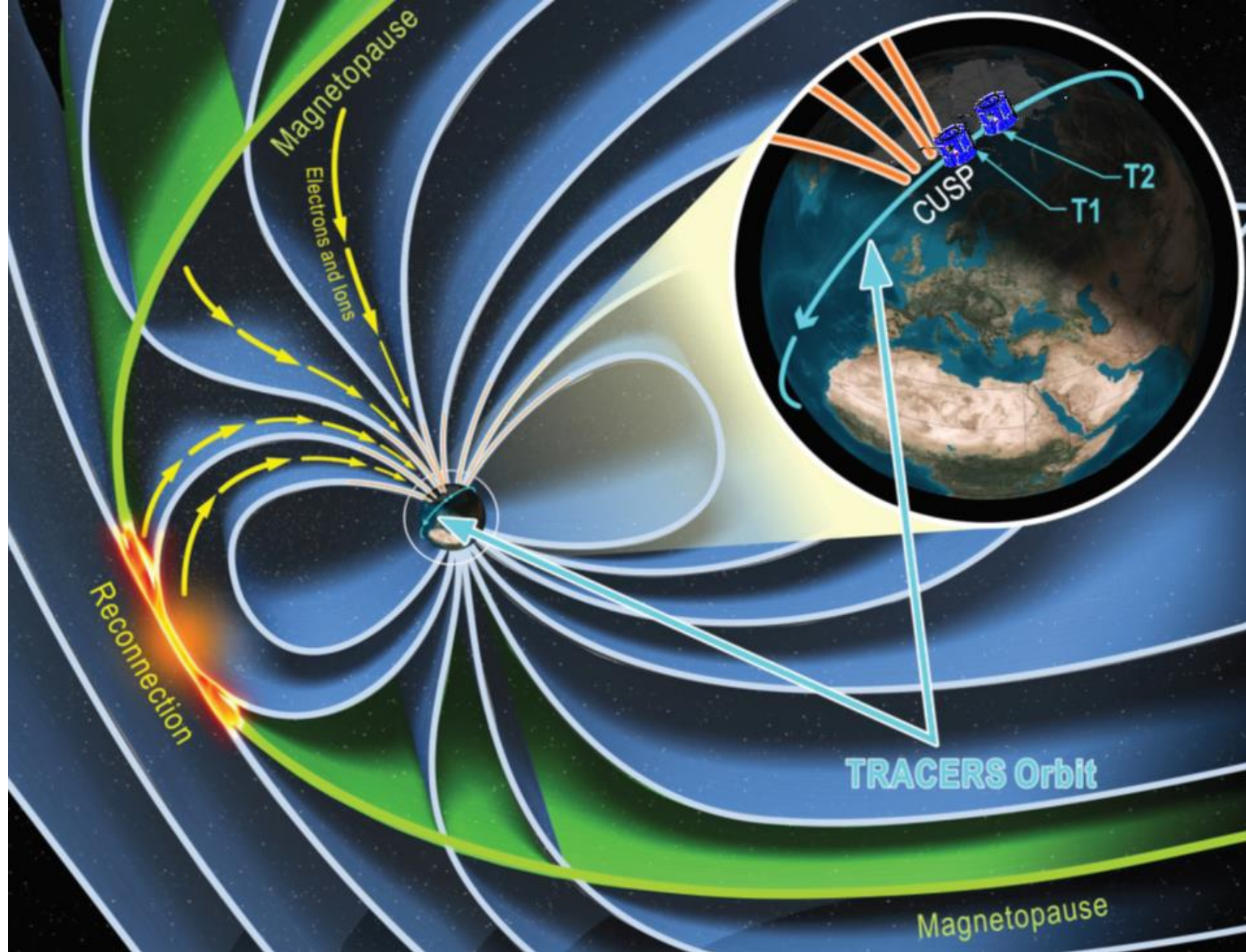
Goals & Objectives

- When, where and why does the solar wind connect to near-Earth space through the Cusp?
- Connecting the Cusp to the Magnetosphere.
- Discovering how spatial or temporal variations in magnetopause reconnection drive Cusp dynamics.



Science Rationale for Mission Design

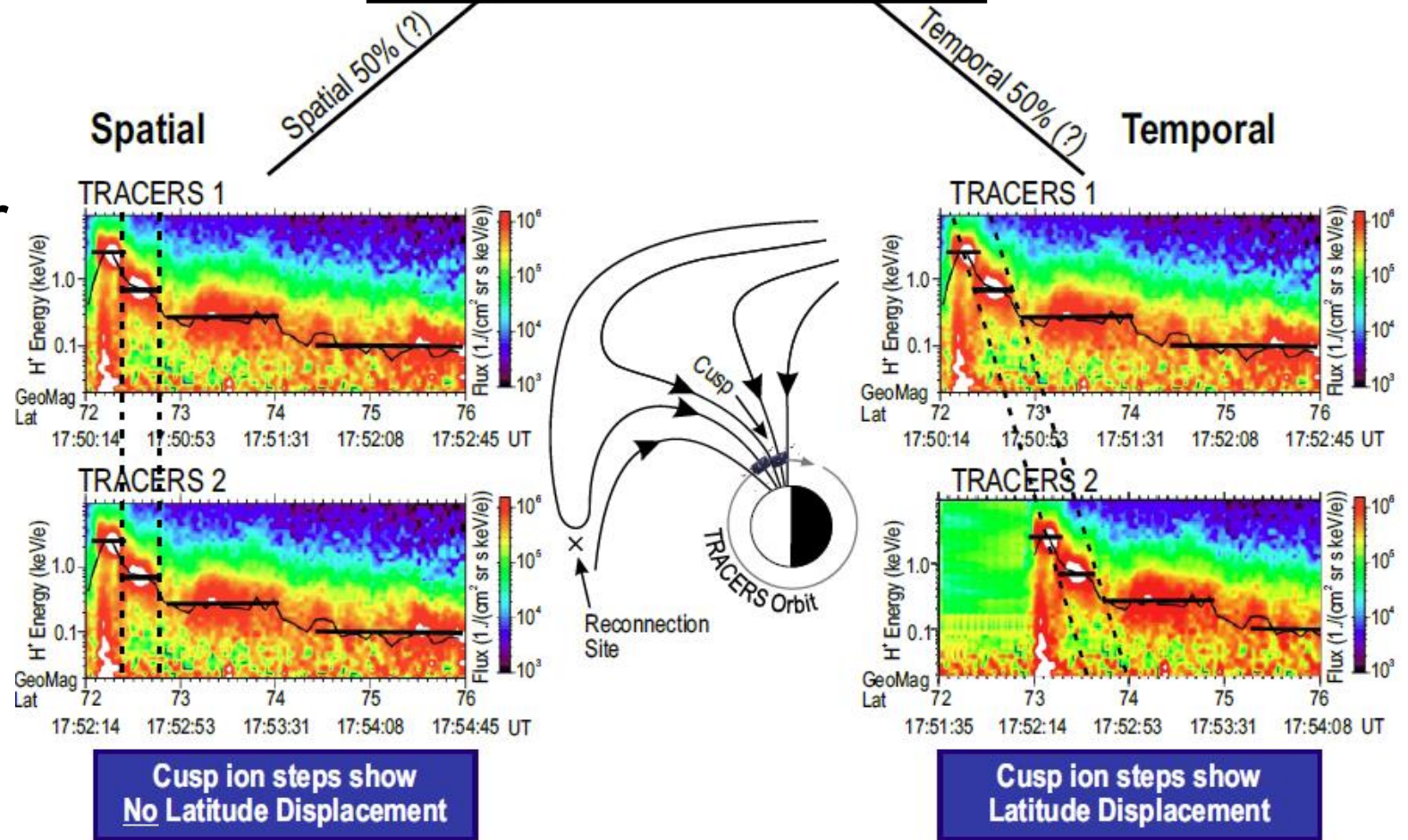
- Two identical satellites to distinguish spatial vs temporal phenomena
- ~600 km sun-sync orbit
- Orbital plane intersects the northern Cusp
- >3000 Cusp crossings in the first year
- 10-120 s separation
- Fields, particles, waves
- 7120.8 magnetometer tech demo (MAGIC)



Science Objective 1

Determine whether magnetopause reconnection is primarily spatially or temporally variable for a range of solar wind conditions.

>3000 CUSP crossings in the first year!

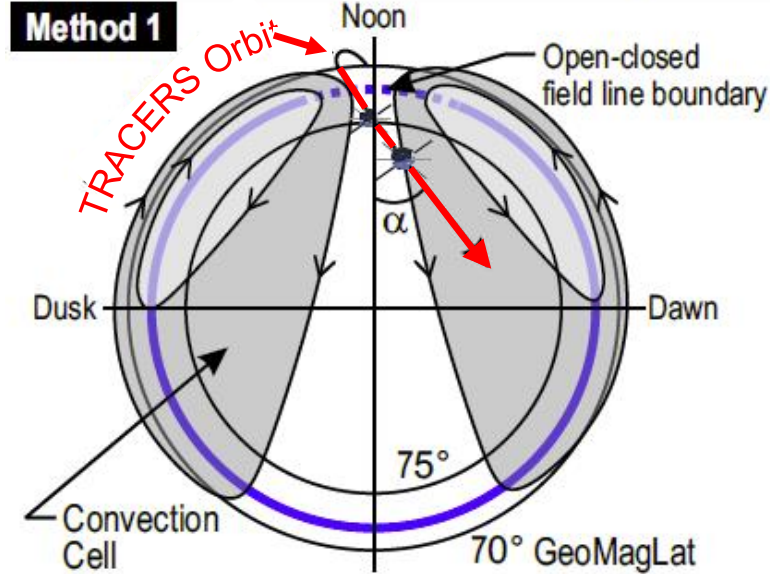


Does Spatial/Temporal dominance depend on Solar Wind conditions?

Science Objective 2

For temporally varying reconnection, determine how the reconnection rate evolves.

Temporal events from Science Objective 1 (Reconnection Rate Measured 2 Ways)



Is V_N Different for TRACERS 1 & TRACERS 2?

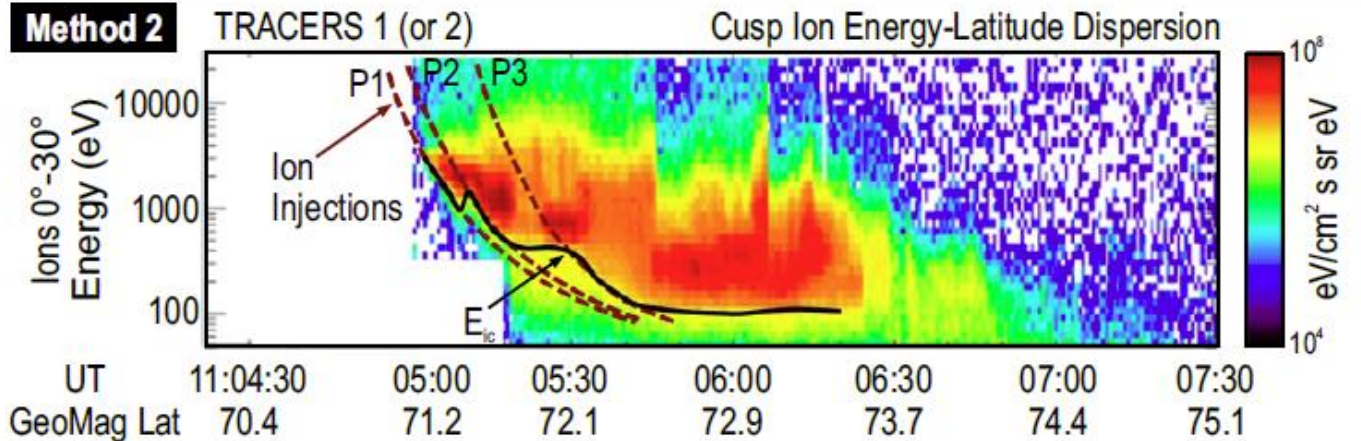
V_C at merging gap = $V_{sc} + V_N + V_M$

V_{sc} = Spacecraft speed

V_N = Reconnection inflow

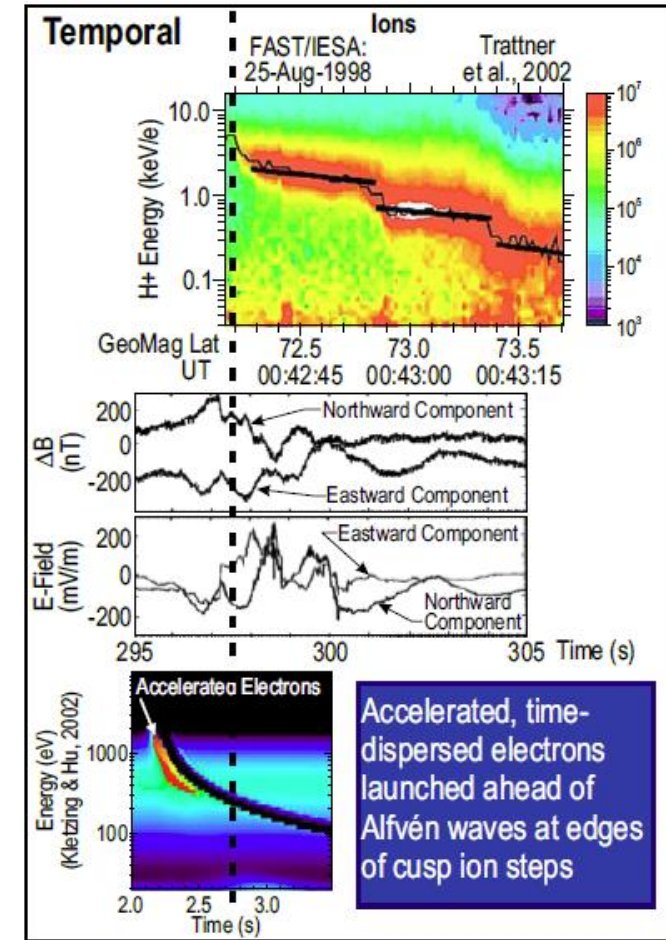
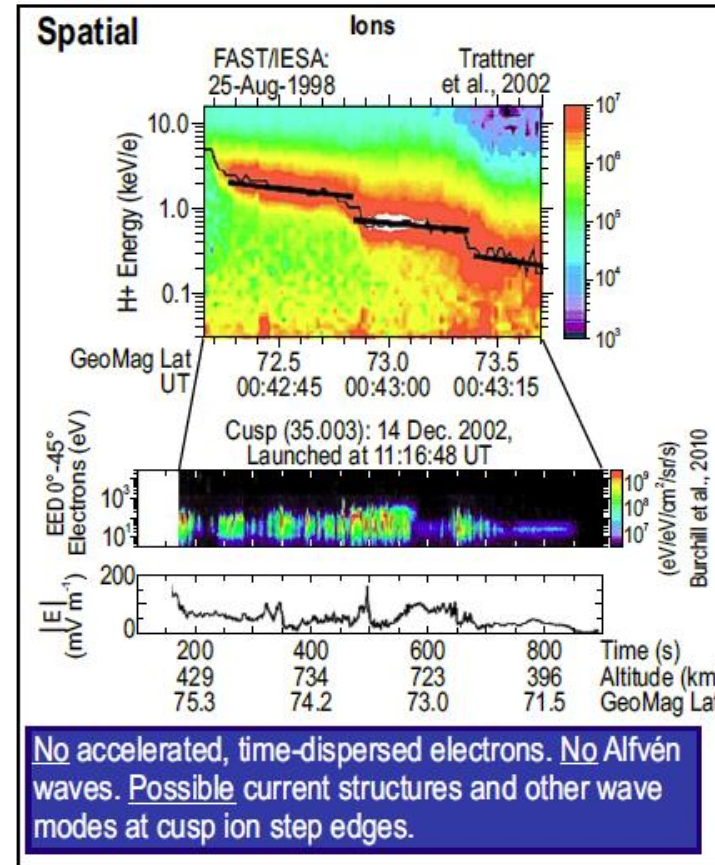
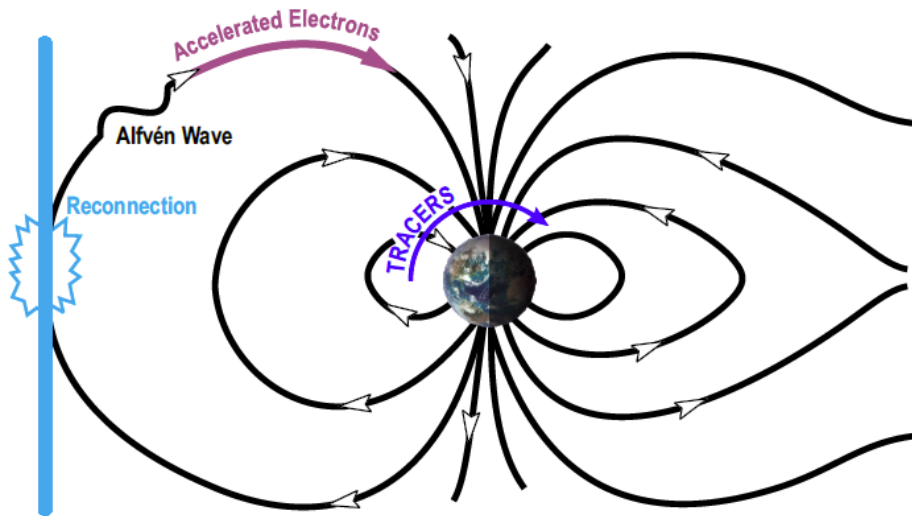
V_M = Motion of open/closed field line boundary from electrons on 2 spacecraft

α = Angle between normal to merging gap and V_{sc} from electric field



Science Objective 3

Determine to what extent dynamic structures in the cusp are associated with temporal versus spatial reconnection.



Mission Science Concept of Operations

Science Region of Interest (ROI)

- ~7 min/orbit during Cusp transit
- High-rate data

Back Orbit (BOR)

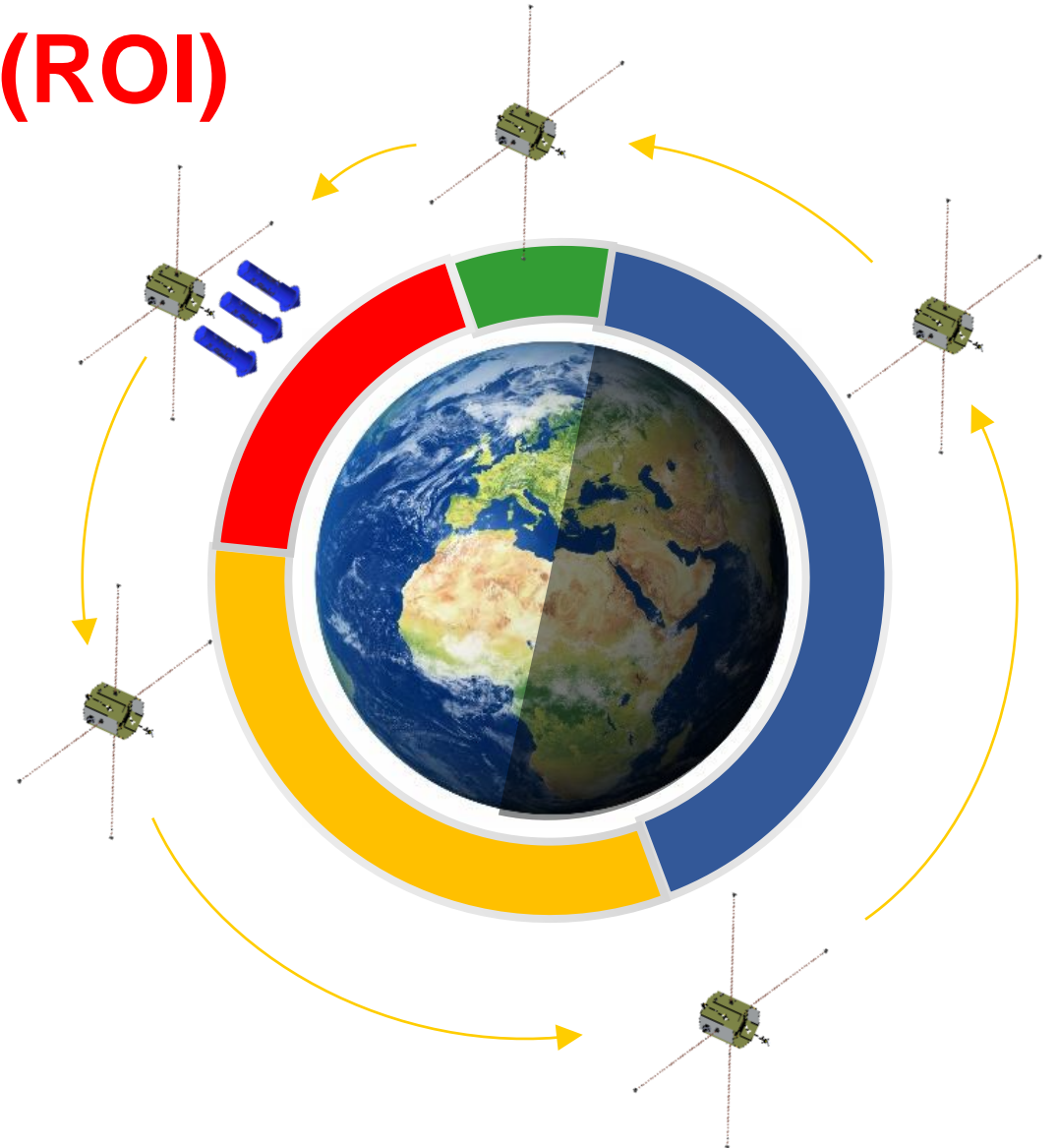
- Once/day alignment of spin-axis
- Housekeeping and low-rate data

Eclipse

- Usually occurs in back orbit
- 35-minute worst-case duration

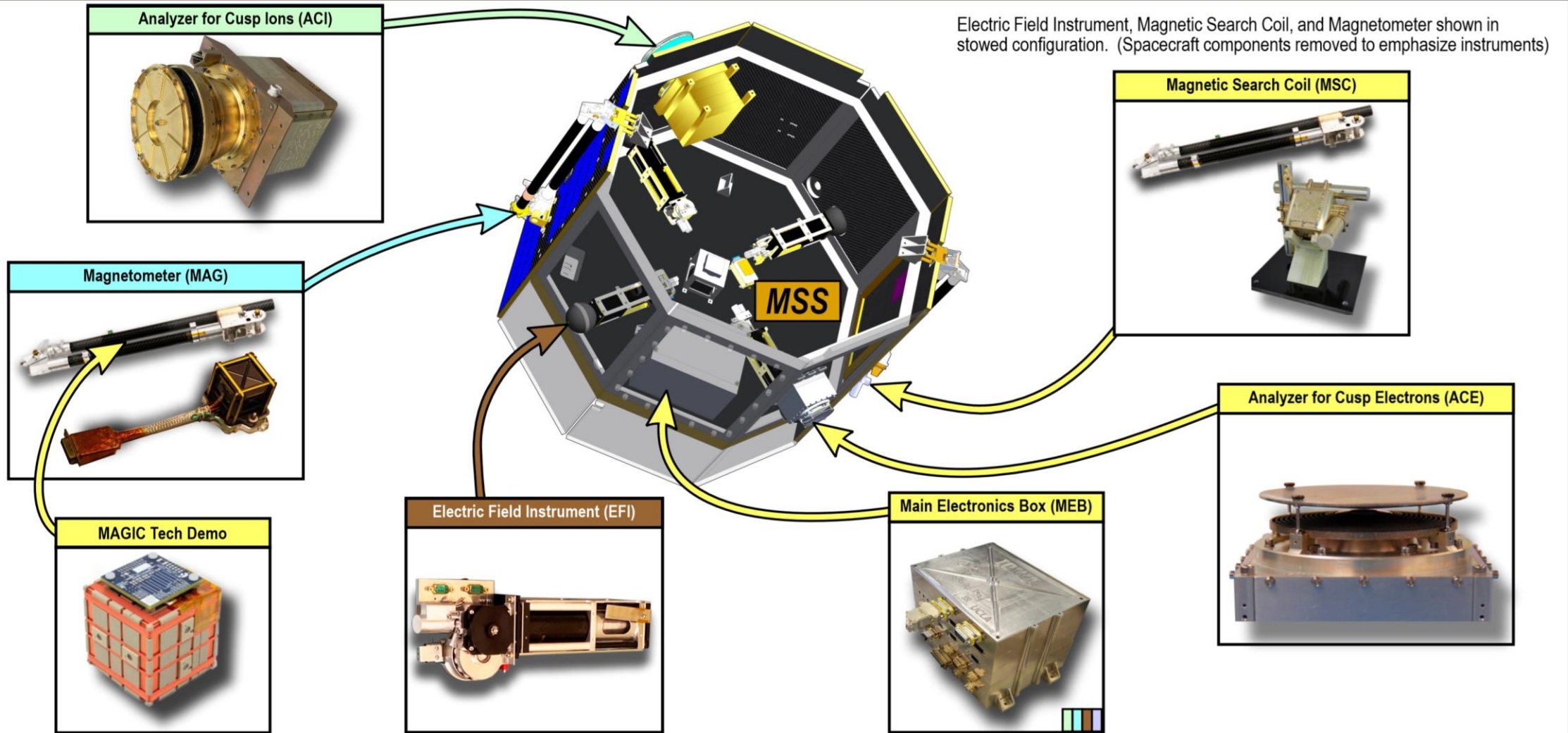
Communications

- Usually in back-orbit
- 14-day forward plan stored commands
- Sequence refreshed every 7 days



TRACERS Instrument Complement

The Two TRACERS Satellites (T1 & T2) Comprise a Spacecraft and an Instrument Suite



TA012232 Phillips

TRACERS Instrument Complement Status

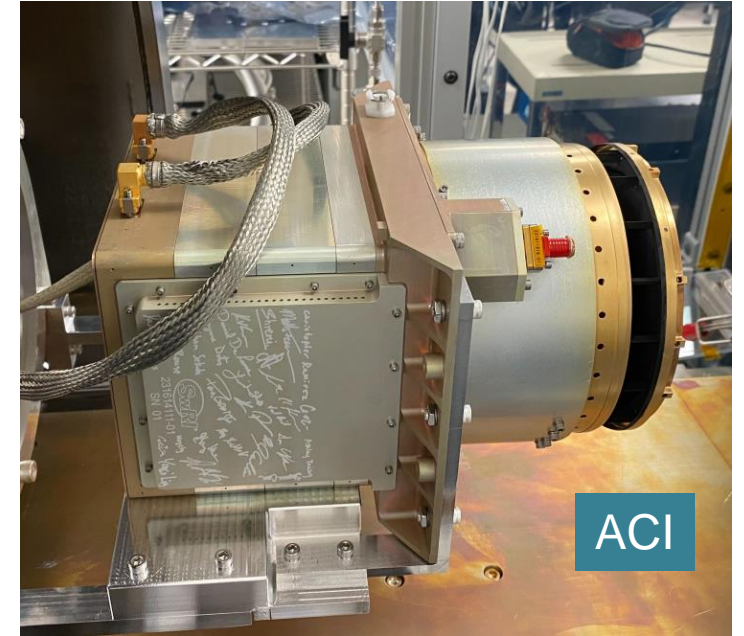
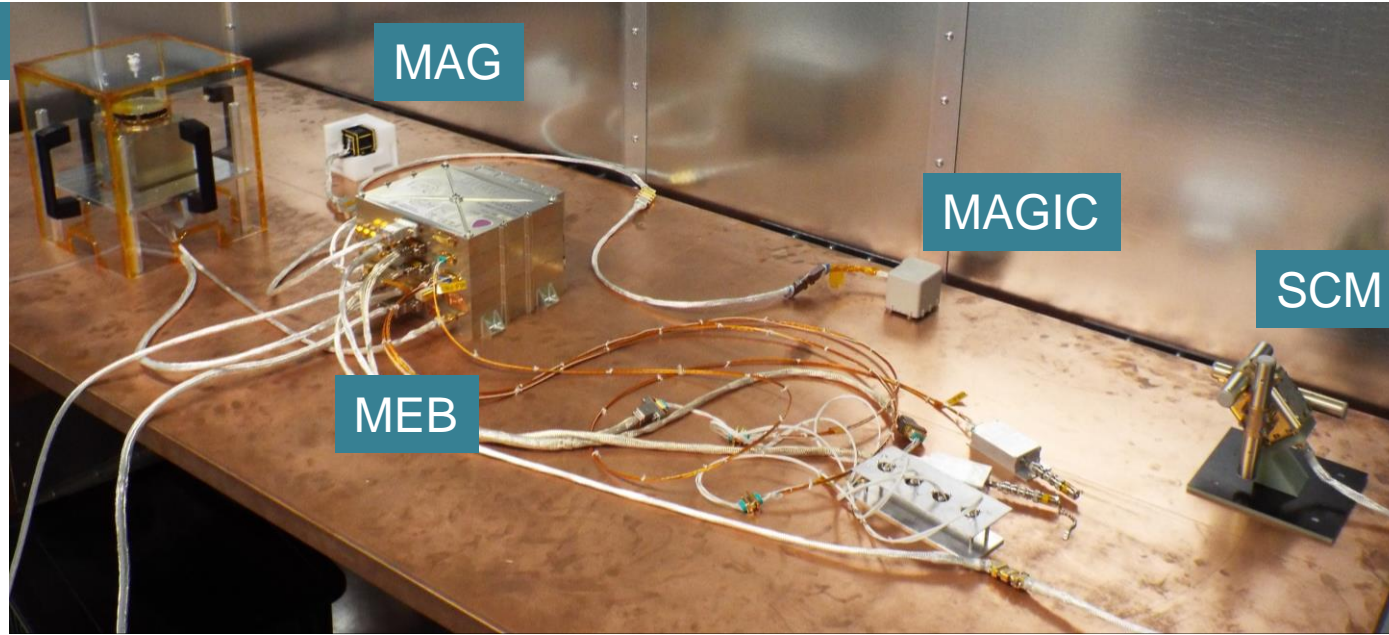
ACE

MAG

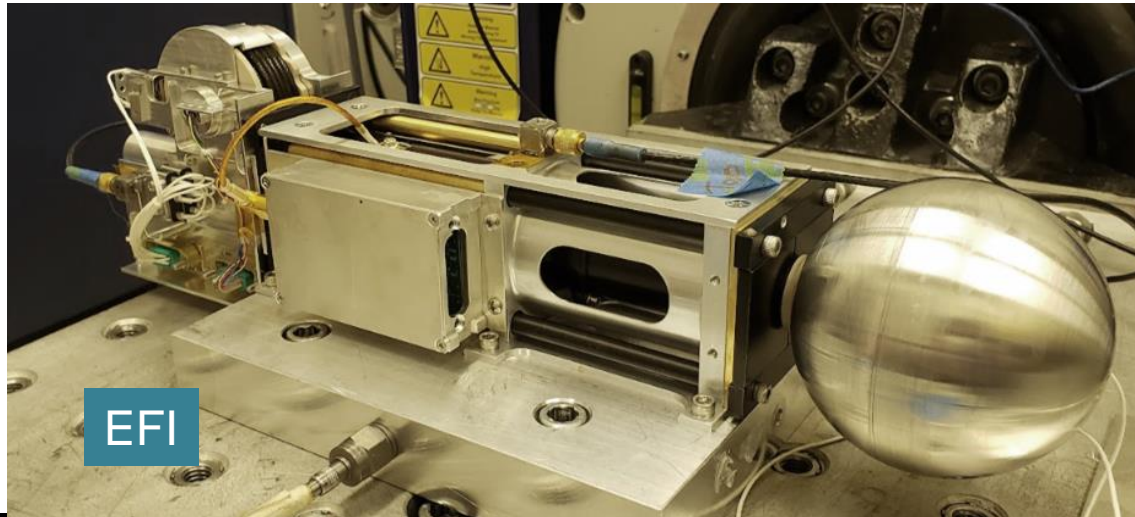
MAGIC

SCM

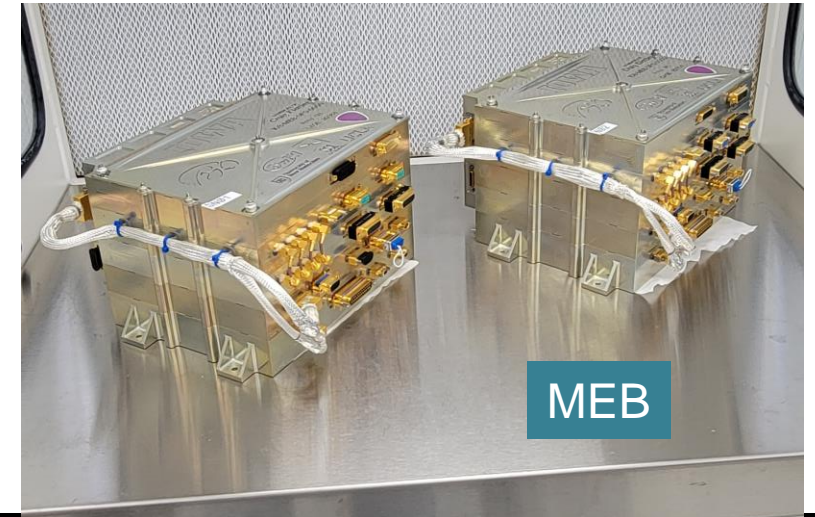
MEB



ACI

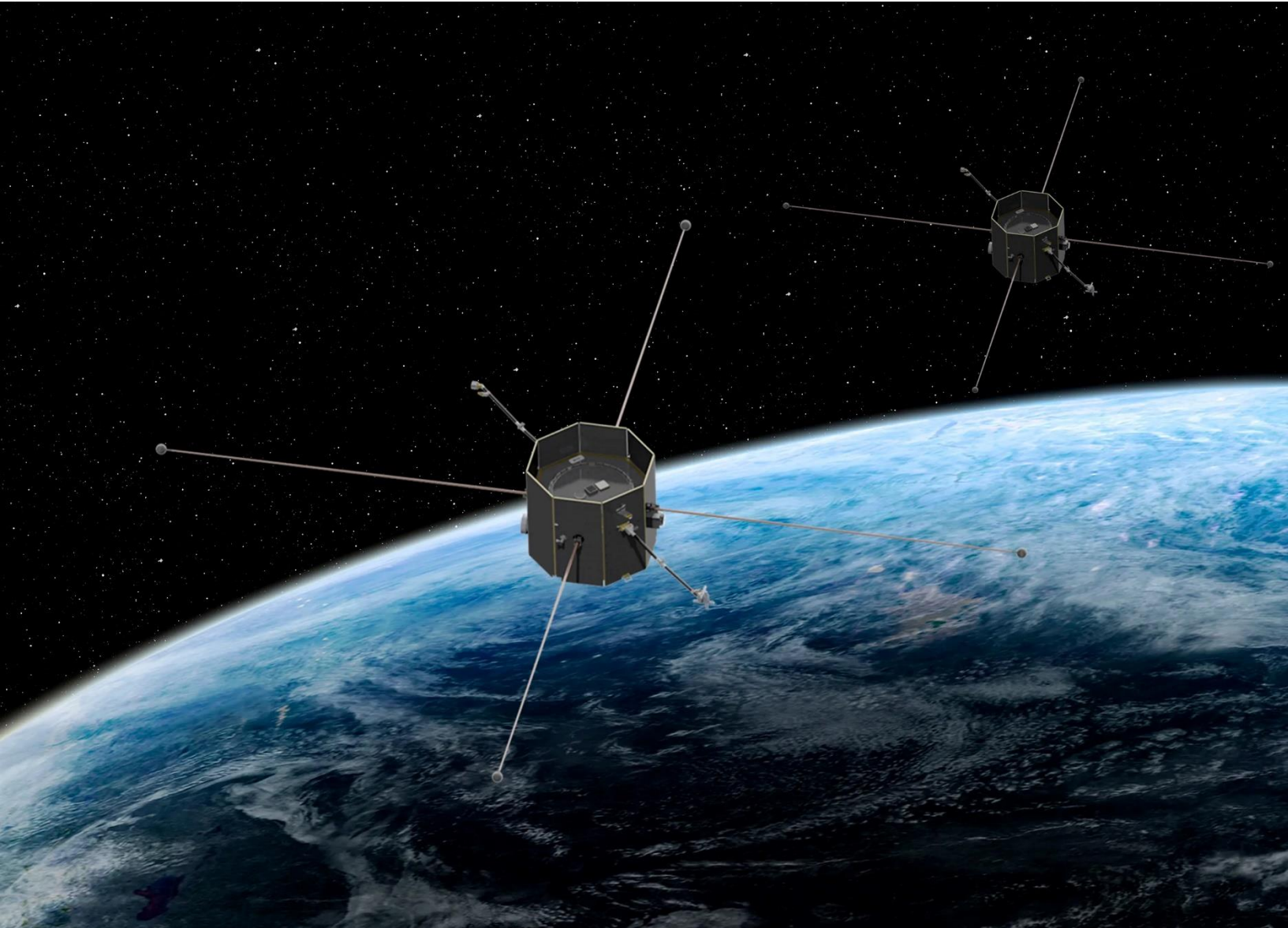


EFI



MEB

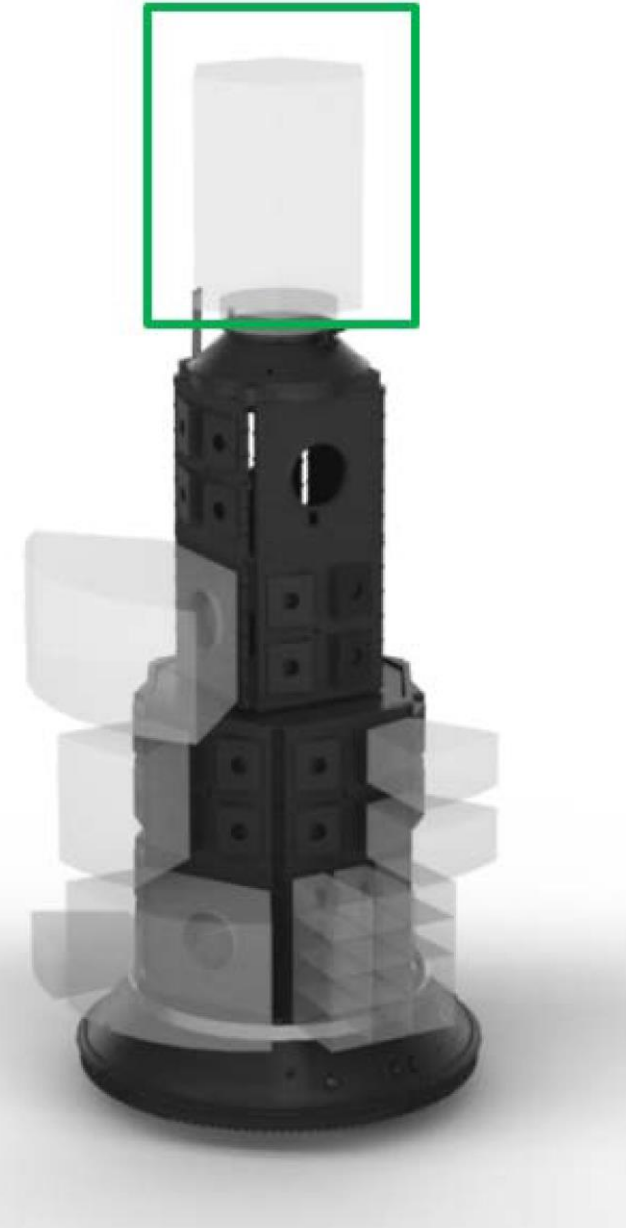
TRACERS Spacecraft

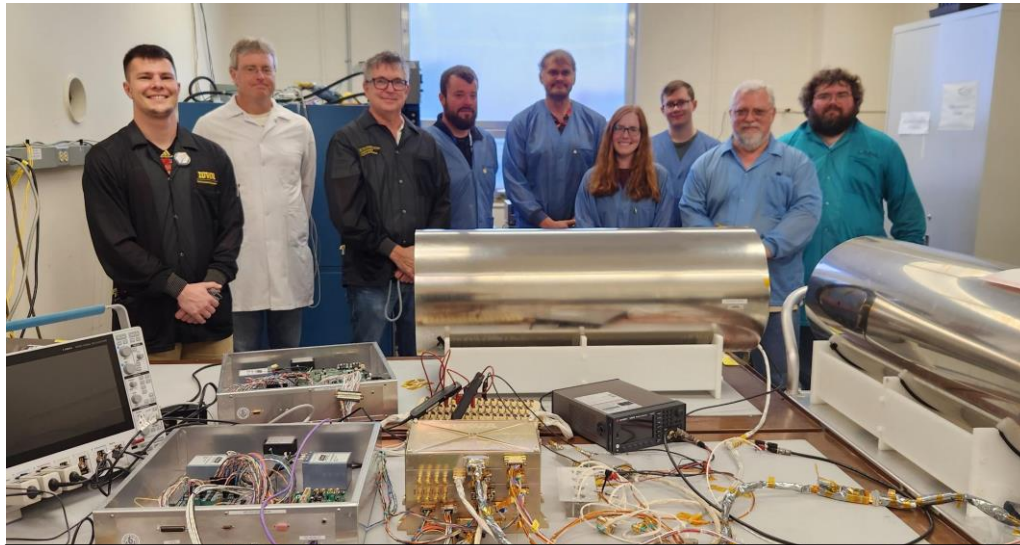


- Two ALTAIR small spacecraft.
- Passively spin stabilized.
- Formation flying.
- Provided by Millennium Space Systems (a Boeing Company).

SpaceX Launch

- On September 29, 2023 NASA announced SpaceX as the Launch Services Provider.
- Primary rideshare (“Cake Topper”) on a Falcon 9.
- Expecting a direct injection to our ideal science orbit.

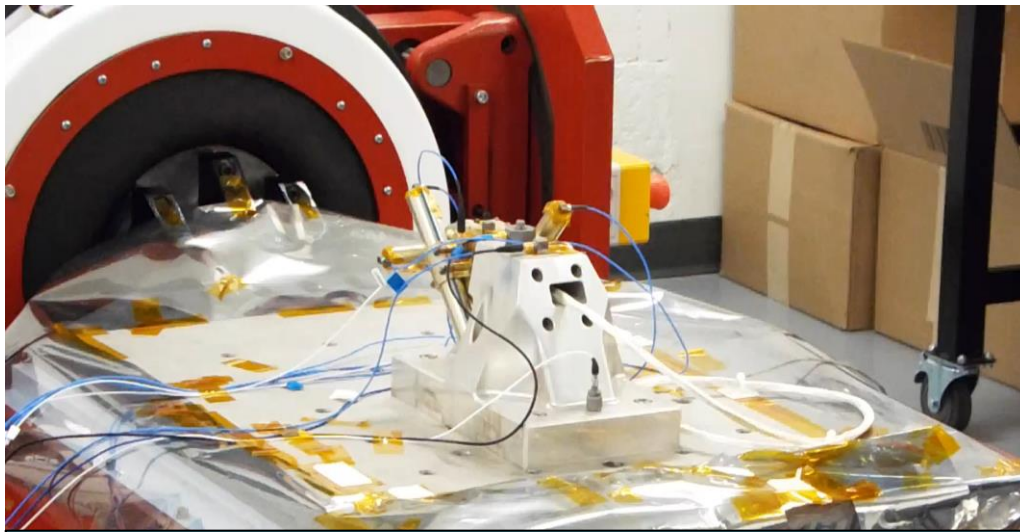




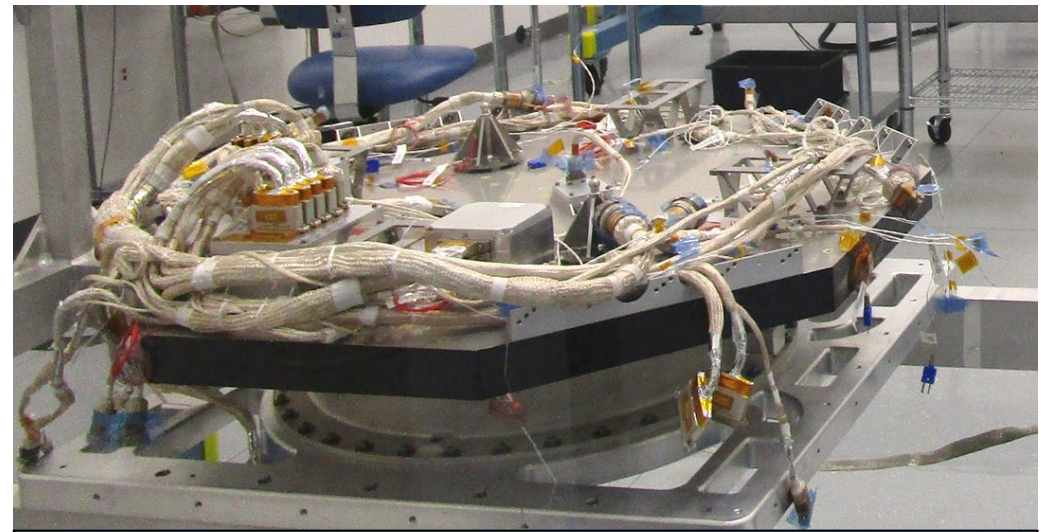
Main Electronics Box (MEB) Integration



Spacecraft Propellant Tank Integration



Instrument Vibration Testing



Spacecraft Deck Awaiting Instruments

TRA-SR (TRACERS student Sounding Rocket)!

- Sounding rocket to be flown in coordination with TRACERS.
- TRACERS-like instruments built and led by graduate and undergraduate students.
- Competed and selected through the Heliophysics Low-Cost Access to Space (H-LCAS) program.
- Led by Prof. Allison Jaynes at Iowa.
- Campaign is managed and run by NASA WSFC.
- Nominal launch window: winter 2025-2026.

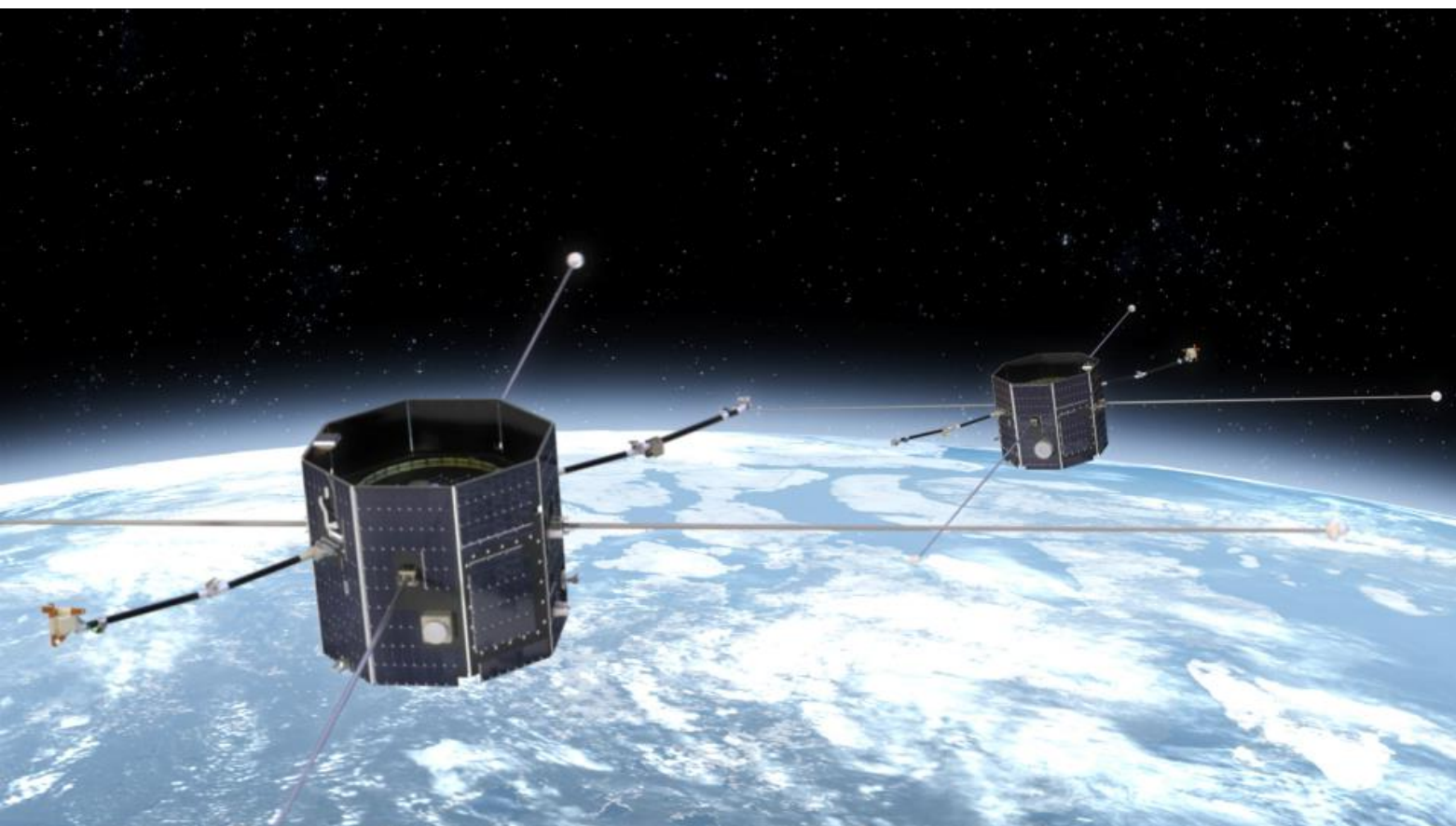
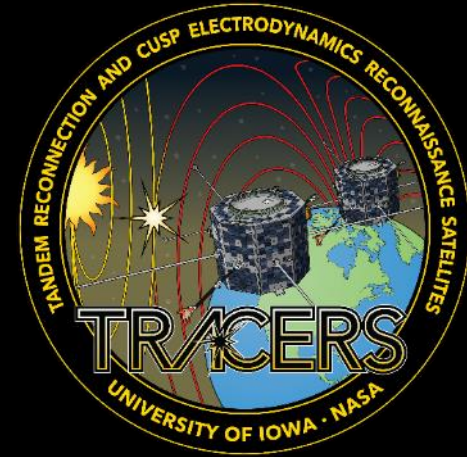


Mission Schedule

Milestone	Date
Instrument Environmental Test	Ongoing
Instruments delivered to Spacecraft 1	June 2024
Instruments delivered to Spacecraft 2	July 2024
Satellite Pre-Environmental Review	August 2024
Operational Readiness Review	December 2024
Launch Readiness Date	April 2025
Start of Operations	May 2025
Public Release of Initial Dataproducts	September 2025
End of Baseline Mission	May 2026
End of Mission	November 2026

TRACERS in April 2025!

- TRACERS will resolve long-standing problems in magnetospheric physics.
- Two identically spacecraft with relatively simple instruments provides new science at low risk.



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Iowa City, Iowa 52242-1479

☎ 319-335-3007

➔ tracers.physics.uiowa.edu

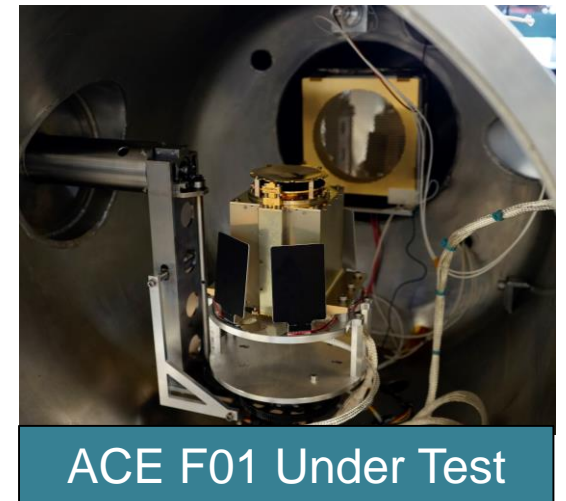
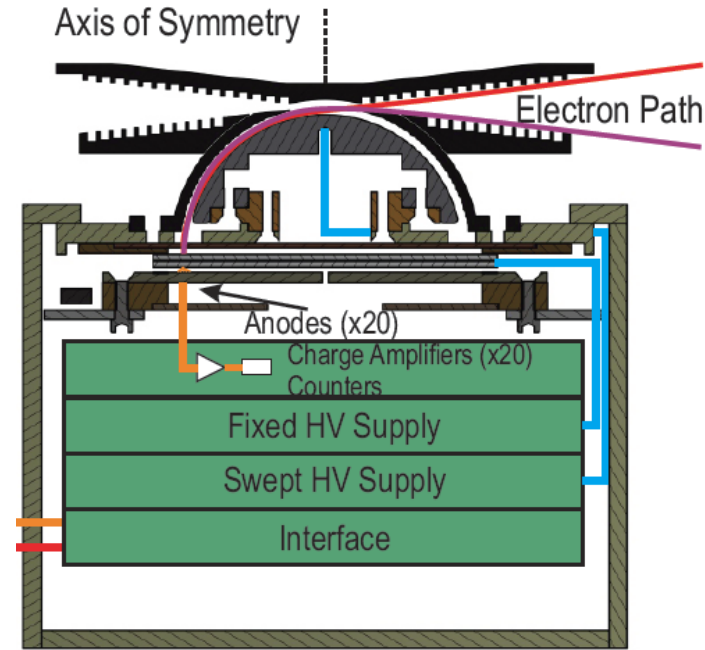
📘 facebook.com/physics.uiowa.edu

📷 @uiowaphysicsastron

✉ david-miles@uiowa.edu

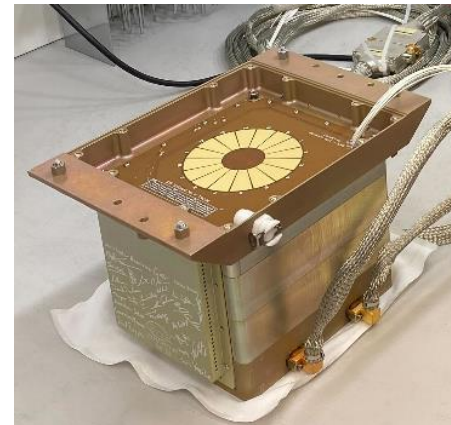
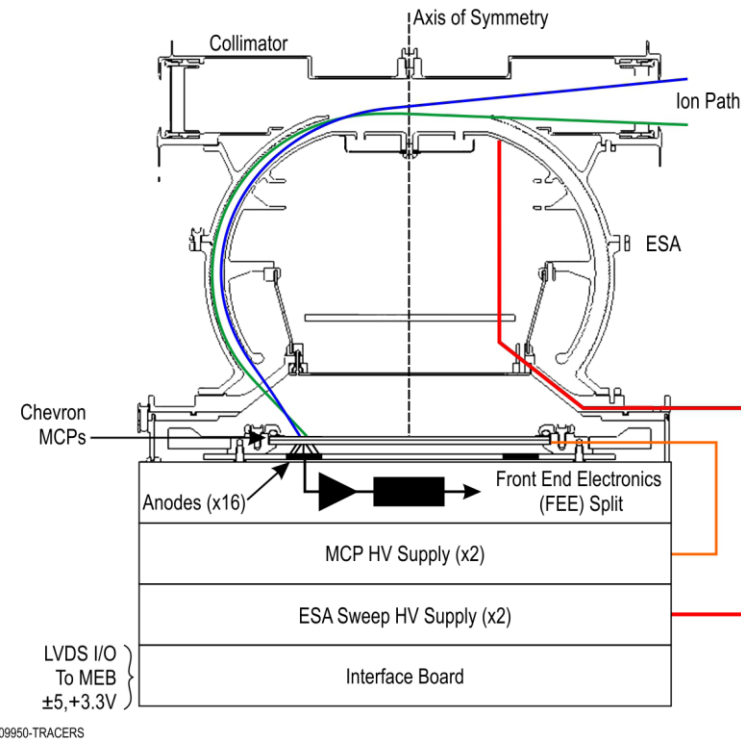
Analyzer for Cusp Electrons (ACE)

- Energy range: 20 eV–15 keV with 15% $\Delta E/E$.
- Angular resolution of $200^\circ \times 7^\circ$ instantaneous and FOV 10° (PA) resolution.
- Per pixel sensitivity of 10^{-4} $\text{cm}^2\text{-sr-eV/eV}$.
- 50 ms time resolution.



Analyzer for Cusp Ions (ACI)

- Energy range: 10 eV–20 keV with 17% $\Delta E/E$.
- Angular resolution of $360^\circ \times 10^\circ$ FOV and 22.5° (PA) resolution.
- Per pixel sensitivity of 3×10^{-4} $\text{cm}^2\text{-sr-eV/eV}$.
- 312 ms time resolution.



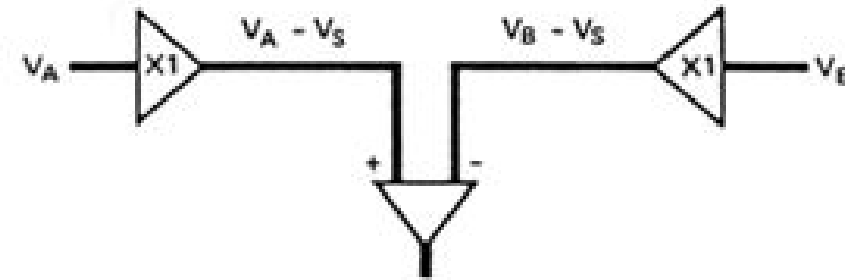
F01 Ebox



F01 ACI in Chamber

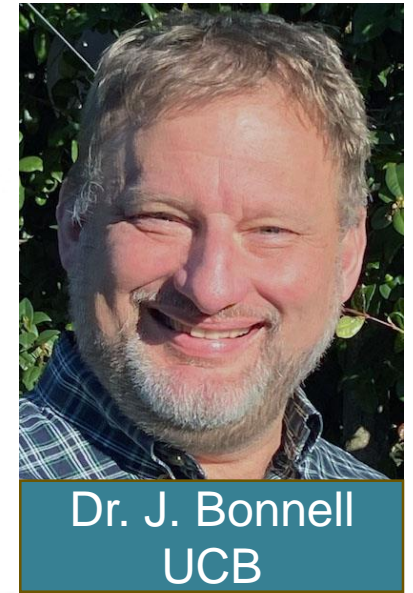
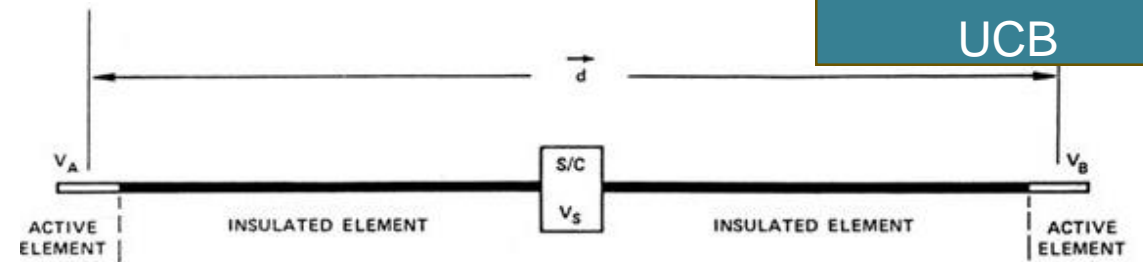
Electric Field Instrument (EFI)

- ± 1 mV/m absolute accuracy (S/C frame).
- $31 \mu\text{V/m}$ resolution (DC E).
- DC-coupled time resolution: 2048 S/s (4V, 2E).
- AC-coupled time resolution: 4 MS/s (1E).
- Four stacer booms.



$$(V_A - V_S) - (V_B - V_S) = V_A - V_B$$

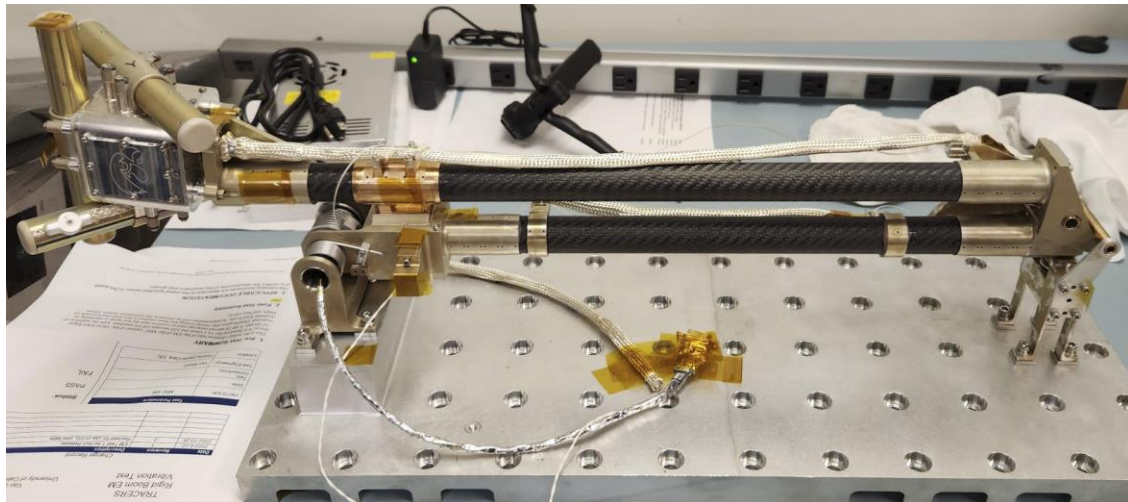
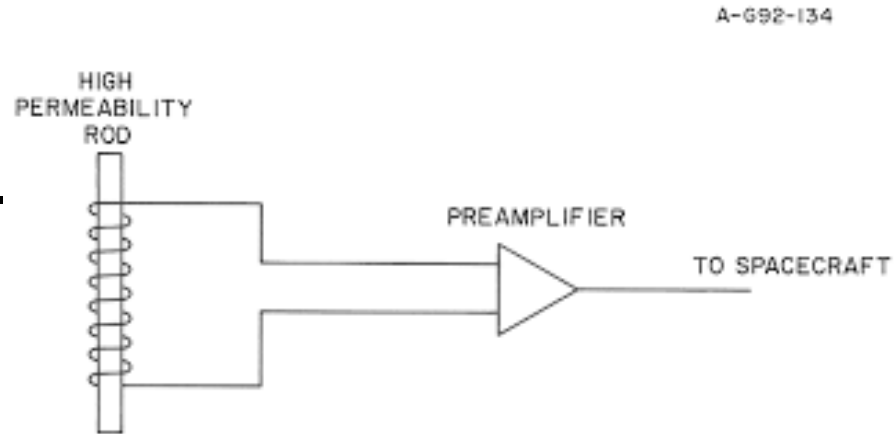
$$(\Phi_1 - \Phi_2) / |d| = (\mathbf{E} + \mathbf{v} \times \mathbf{B}) \cdot \mathbf{d}$$



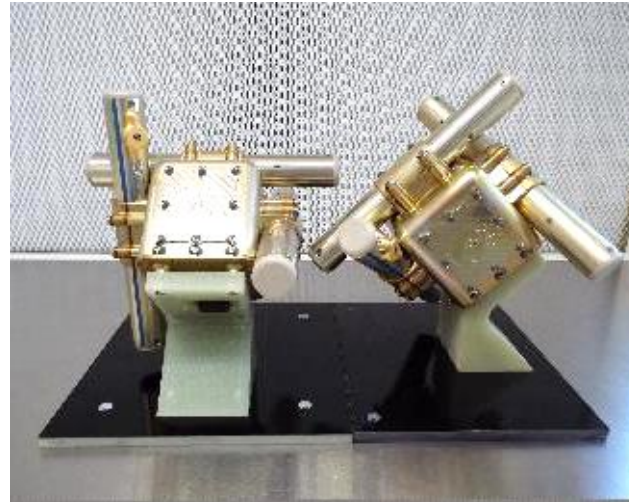
EFI Stacers

Magnetic Search Coil (MSC)

- Frequency: ~1 Hz to 1024 Hz.
- Spectral: 10^{-7} nT @ 10 Hz.
- Spectral: 10^{-9} nT @ 100 Hz.
- Boom mounted.



MSC/MAG EM Rigid Booms



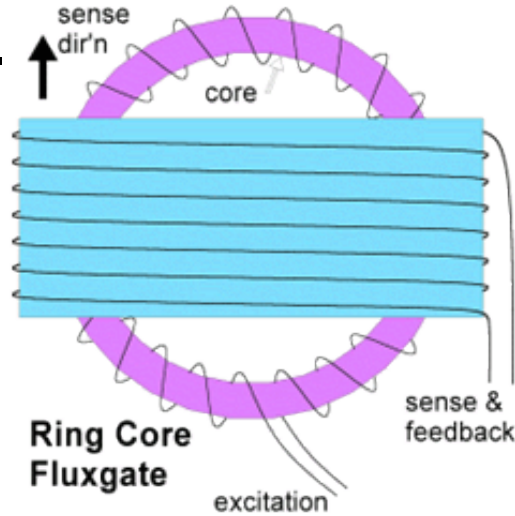
MSC F01/F02



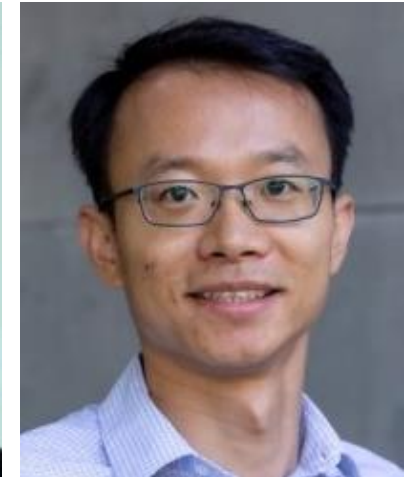
MSC F01+EFI F01

Fluxgate Magnetometer (MAG)

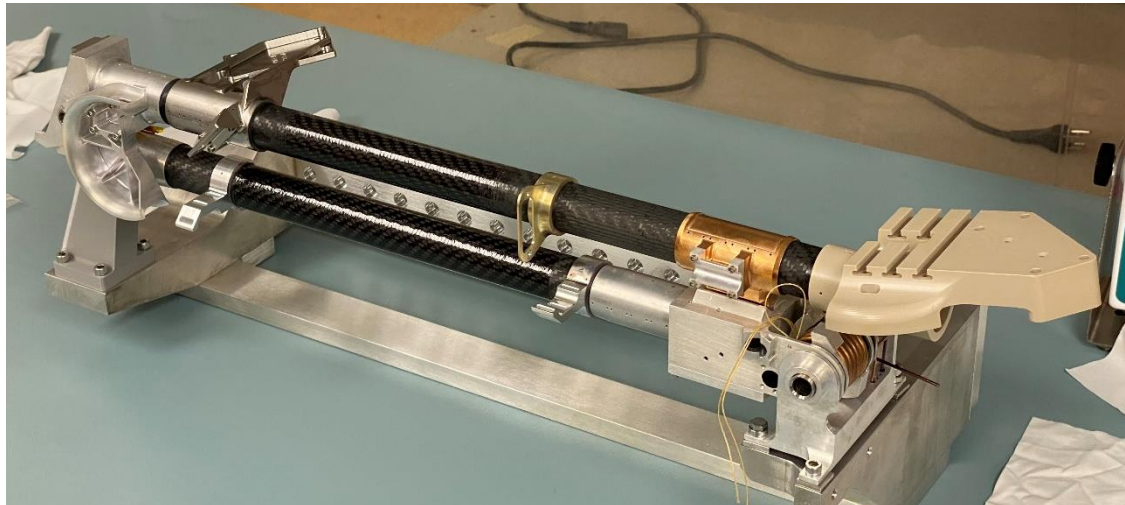
- Dynamic range: $\pm 60,000$ nT.
- Resolution: 0.5 nT.
- Accuracy: 10 nT.
- Boom mounted.



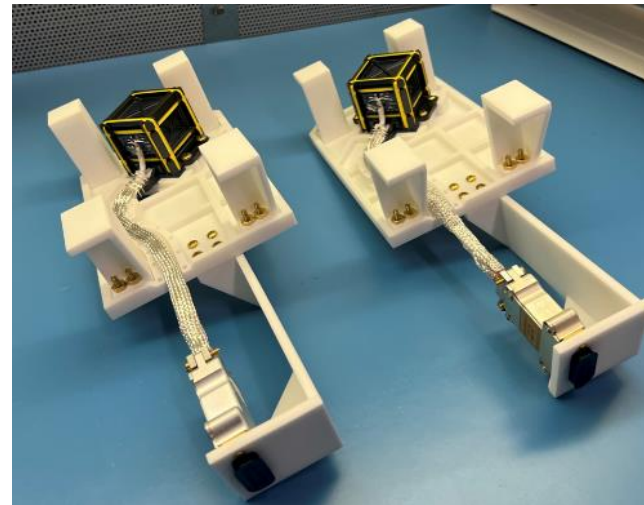
Dr. R. Strangeway
UCLA



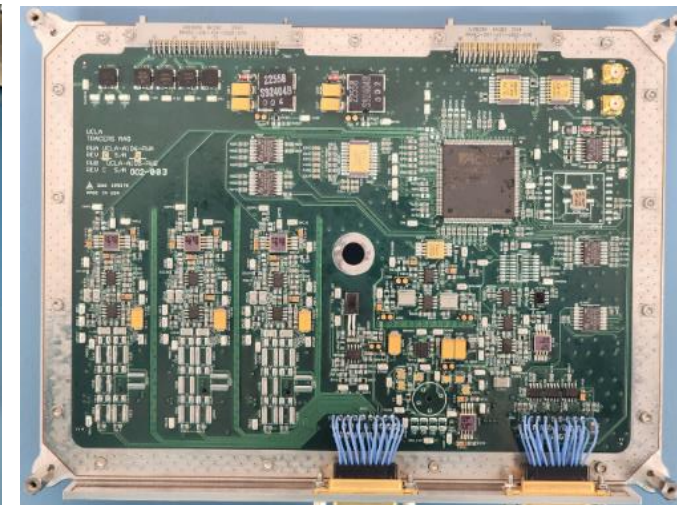
Prof. H. Cao
UCLA



MSC/MAG EM Rigid Booms



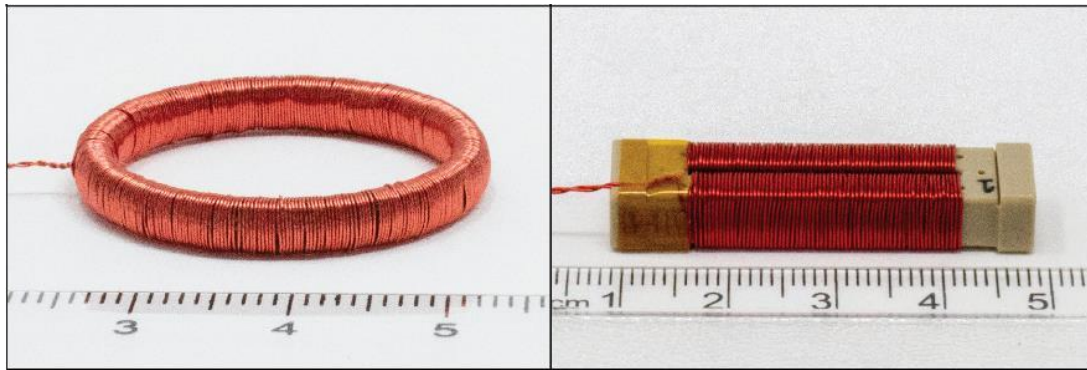
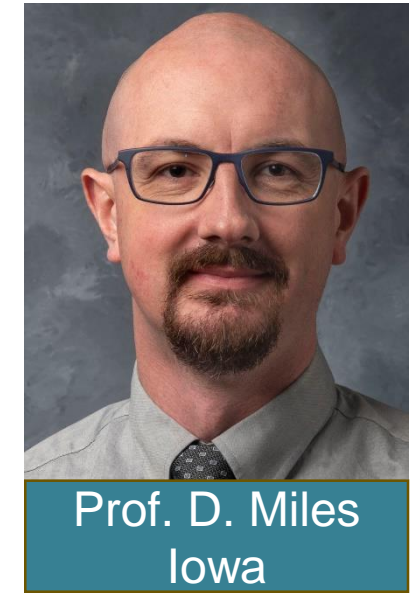
Sensors F01 and F02



Flight Board S/N2

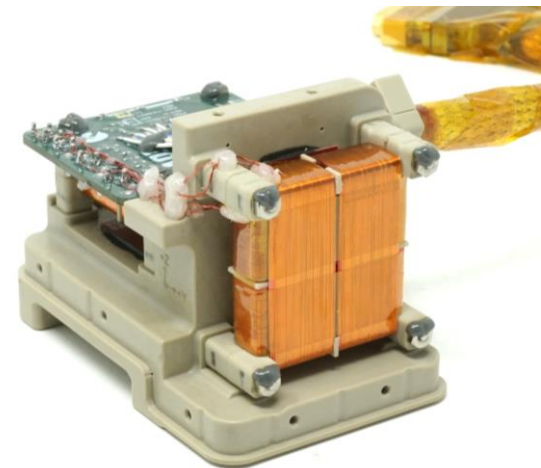
MAGIC NPR 7120.8 Tech Demo

- Low-noise fluxgate cores (5 pT/ $\sqrt{\text{Hz}}$ at 1 Hz).
- Heritage ringcore sensor.
- New Tesseract sensor.
- Boom mounted.

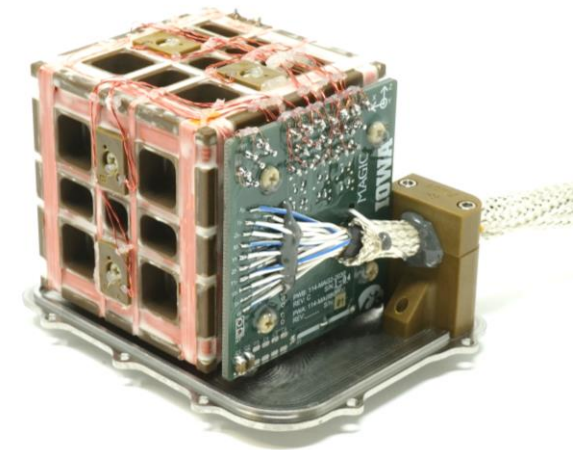


Ringcore

Racetrack Core



Ringcore Sensor



Tesseract Sensor

Simulation and Science Team

- Context for TRACERS data from global hybrid simulations.
- Field line tracing allows direct comparison with TRACERS particle measurements.
- Focused science team with relevant domain expertise.



Prof. Y. Lin
Auburn



Dr. S. Fuselier
SwRI



Prof. K. Goodrich
WVU

